

# Investigation of the Aerodynamics Performance of Horizontal Axis Wind Turbine Rotor

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**Abstract** – In order to convert wind energy to mechanical energy, the design of the axis wind turbine is very crucial, the rotor is an important component of the wind turbine. This study investigates the aerodynamics performance of horizontal axis wind turbine rotor. By obtaining the power coefficient, we found that about forty percent of the kinetic energy of the undisturbed wind flowing to the swept area of the rotor can be extracted as useful power. Also, we found that the twisting of the blades of a horizontal axis wind turbine has a positive impact on the performance of its rotor, as twisting optimize angle of attack and maximize the lift force. Other factors such as high lift-to-drag ratio, high maximum lift coefficient and high tip speed ratio were found to bring about improvement to the performance of a horizontal axis wind turbine.

**Index Terms** – Wind turbine, aerodynamics, horizontal wind turbine, turbine blade

## 1 INTRODUCTION

Wind energy has risen to become one of the most applied form of renewable energy [1]–[5]. Different processes and plants utilizes wind energy [6]–[17]. Horizontal axis wind turbine is the most commercially available and has gained the attention of investors [18].

The aerodynamic performance of an Horizontal Axis wind turbine rotor depends largely on design parameters including number of blades, airfoil shapes, blade twist along the rotor radius, blade pitch, tip speed ratio and rotor hub height (or tower height) [19].

In addition, the aerodynamic performance of a Horizontal Axis Wind Turbine rotor can be evaluated using: Blade Element Theory and Momentum Theory [19]. The blade element theory assumes that each blade of a horizontal axis wind turbine can be divided into many small “elements” that are independent of other elements and operates aerodynamically as 2D air-foils [19]. Also, the blade theory is based on two main assumptions: No aerodynamic interactions exist between the different blade elements and the lift and drag coefficients solely determine the forces on the blade elements [20].

On the other hand, the momentum theory considers momentum loss through the plane of the rotor based on work done by the wind [19]. The combination of “blade element” and “momentum” theories give arise to the Blade Element Momentum (BEM) theory [19]. The “BEM” theory is used to determine the total thrust and thrust on a horizontal axis wind turbine blade through an iterative process- which is carried out for each blade element [19]. The iterative process for each blade element is as follows:

1. Initialize the tangential induction factor,  $a'$ , to be zero and the axial induction factor,  $a$ , to be any value.
2. Determine the inflow angle,  $\phi$ , using:  
$$\phi = \tan^{-1} \left[ \frac{U_{\infty}(1-a)}{\Omega r(1+a')} \right]$$
3. Determine the angle of attack “ $\alpha$ ” for the airfoil:  $\alpha = \phi - \beta$ .
4. Check for the drag and lift coefficient,  $C_D$  and  $C_L$  from the performance dataset gotten from XFOIL or XFLR5 program.
5. Determine the thrust and torque contribution from the annulus using Blade Element Theory:

Total thrust “ $dT$ ” for “ $B$ ” number of blades is given as:

$$dT = \frac{B}{2} \rho V^2_{total} [C_L \cos \phi + C_D \sin \phi] c dr.$$

Also, the total torque “ $dQ$ ” for “ $B$ ” number of blades is given as:

$$dQ = \frac{B}{2} \rho V^2_{total} [C_L \sin \phi - C_D \cos \phi] c r dr.$$

6. Update the values of axial induction factor “ $a$ ” and tangential induction factor “ $a'$ ”, using the momentum theory. The following equations are used to update the values of the two induction factors:  
Thrust extracted by each rotor annulus,  $dT = 4\pi r \rho U_{\infty}^2 (1-a) a dr$   
Torque extracted by each rotor annulus,  $dQ = 4\pi r^3 \rho U_{\infty} \Omega (1-a) a' dr$ .
7. Repeat the above procedures from step 2 to step 6 until the values of “ $a$ ” and “ $a'$ ” converge.
8. The above calculation is done for each element. Thus, we can sum up the thrust and torque of all the elements to obtain the rotor thrust and torque. Finally, the power

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generated by the rotor and its power coefficient are calculated.

### 1.1 Estimation of Reynold's Number and The Local Total Velocity

The given parameters in the question include:

Rotor radius 'R'=6m,

Diameter of Rotor 'D'=2R=2×6=12m

Length of blade, L=5m

Airfoil adopted = "NACA 0020"

Chord Length 'c'=1m

Angular velocity of rotor ' $\Omega$ '=6rad/s

Natural wind speed ' $U_\infty$ '=8m/s

Density of air ' $\rho$ '=1.2kg/m<sup>3</sup>

Viscosity of air ' $\mu$ '=1.8×10<sup>-5</sup>kgm<sup>-1</sup>s<sup>-1</sup>

- a. To calculate local total velocity at mid-span of the blade " $V_{total}$ ", the diagram below is considered:

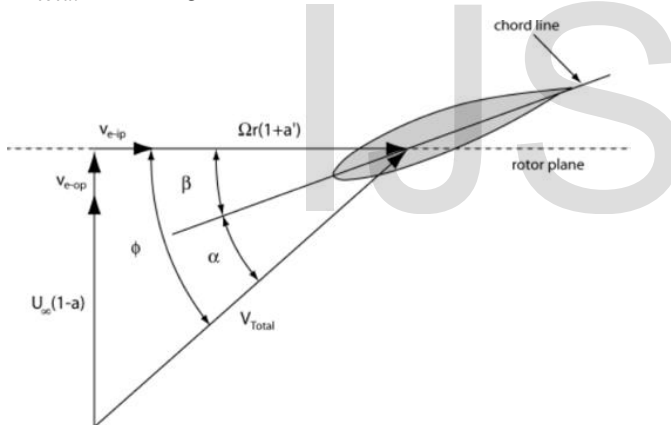


Figure 1: Vector representation of the velocity components of a wind turbine blade [19].

From Pythagoras' theorem:

$$V^2_{total} = [\Omega r(1 + a')]^2 + [U_\infty(1 + a)]^2$$

Where axial induction factor "a" and tangential induction factor "a'" equal to zero

Thus,  $V_{total}$  becomes:

$$V^2_{total} = (\Omega r)^2 + U_\infty^2 \text{ where } r = 3.5\text{m (i.e. radius at mid-span of the blade)}$$

Recall, Linear velocity of rotor rotation at mid-span  $v = \Omega r = 6 \times 3.5 = 21\text{m/s}$

$$V^2_{total} = 21^2 + 8^2 = 441 + 64 = 505$$

$$V_{total} = \sqrt{505} = 22.5\text{m/s} \approx 23\text{m/s}$$

Therefore, the total local velocity at mid-span of the blade,  $V_{total}$  is 23m/s.

Representative Reynold's number for the rotating blades 'Re':

$$Re = \frac{\text{Density of air} \times \text{Total local velocity} \times \text{Characteristic length}}{\text{Viscosity of air}} = \frac{\rho \times V_{total} \times c}{\mu}$$

Where characteristic length = chord length = 'c'

$$Re = \frac{1.2 \times 23 \times 1}{1.8 \times 10^{-5}} = 1,533,333 = 1.5 \times 10^6$$

Therefore, the Reynold's number for the rotating blades is  $1.5 \times 10^6$ .

### 1.2 Aerodynamic Coefficients and X-foil Program Execution

#### 1.2.1. Definition of Aerodynamic Coefficients

The definitions of lift and drag coefficients are as shown below:

Lift coefficient is a dimensionless parameter that shows the relationship between the lift force ' $F_L$ ' acting on a 2D airfoil, density of air ' $\rho$ ', wind speed ' $U$ ' and reference area ' $A$ '.

$$\text{Lift Coefficient, } C_L = \frac{F_L}{\frac{1}{2} \rho U^2 A}$$

Drag coefficient is a dimensionless parameter that indicates the relationship between the drag force ' $F_D$ ' acting on a 2D airfoil, density of air ' $\rho$ ', wind speed ' $U$ ' and reference area ' $A$ '.

$$\text{Drag Coefficient, } C_D = \frac{F_D}{\frac{1}{2} \rho U^2 A}$$

where ' $F_L$ ' is the lift force (in Newton), ' $F_D$ ' is the drag force (in Newton),  $\rho$  is the density of air (in Kg/m<sup>3</sup>), ' $A$ ' is the reference area and ' $U$ ' is the upstream wind velocity.

#### 1.2.2. X-Foil Program Execution and Results

The following parameters were used in the X-FOIL program to get the relationship between various parameters:

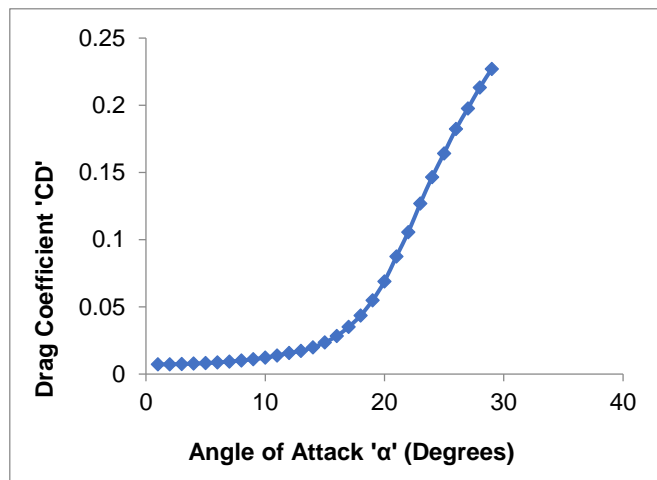
Reynold's Number  $Re = 1,500,000$

Range of angles of attack:  $0^\circ < \alpha < 30^\circ$  (i.e. ' $\alpha'$ ' lies between 1 and 29, with both numbers inclusive)

Angle of attack increment,  $\Delta\alpha = 1^\circ$  NACA 0020 was adopted  
 The table below shows the values of lift coefficient ' $C_L$ ', drag coefficient ' $C_D$ ' and angle of attack ' $\alpha$ ' exported from the X-foil program:

**Table 1: Variation of angle of attack 'α' with lift coefficient 'C<sub>L</sub>' and drag coefficient 'C<sub>D</sub>'.**

Angle of Attack, α	Lift Coefficients, C <sub>L</sub>	Drag Coefficient, C <sub>D</sub>
1	0.1103	0.00722
2	0.2202	0.00731
3	0.3292	0.0075
4	0.437	0.0078
5	0.5436	0.00817
6	0.6481	0.00868
7	0.7499	0.00933
8	0.8477	0.01012
9	0.9383	0.01106
10	1.0233	0.01224
11	1.1403	0.01382
12	1.2562	0.01567
13	1.2814	0.01731
14	1.338	0.01992
15	1.3908	0.02351
16	1.4345	0.02834
17	1.4633	0.03509
18	1.4827	0.04354
19	1.4812	0.05488
20	1.4624	0.06895
21	1.4132	0.08756
22	1.3743	0.1056
23	1.3179	0.012688
24	1.2775	0.14661
25	1.2551	0.16422
26	1.2319	0.18241
27	1.2283	0.19762
28	1.2231	0.2133
29	1.2265	0.22707



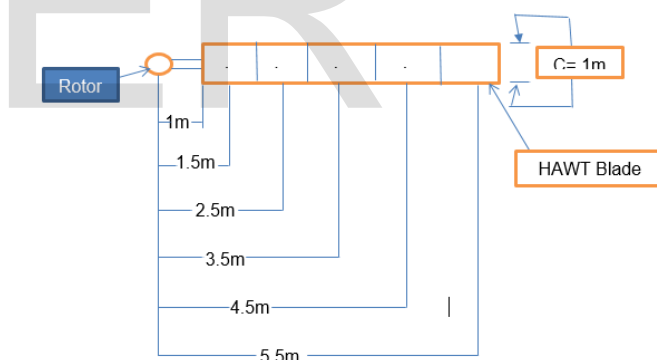
**Figure 3 Variation of drag coefficient with respect to angle of attack**

**1.3 Blade Element Momentum Theory**

Pitch angle 'β'=8°,

Tangential induction factor a'=0

Lift coefficient 'C<sub>L</sub>'=1.25 Drag coefficient 'C<sub>D</sub>'= 0.24+0.015(α-30)



**Figure 4 Distance of blade elements from the rotor centre.**

From the figure above, the blade is divided into five elements. Also, It is clearly seen from the figure that r<sub>1</sub>=1.5m, r<sub>2</sub>=2.5m, r<sub>3</sub>=3.5m, r<sub>4</sub>=4.5m and r<sub>5</sub>=5.5m. These values were used in the iteration process discussed in section 3.1. using BEM theory and a MATLAB code was written to perform the iteration process. This MATLAB code can be viewed in the appendix section of this report.

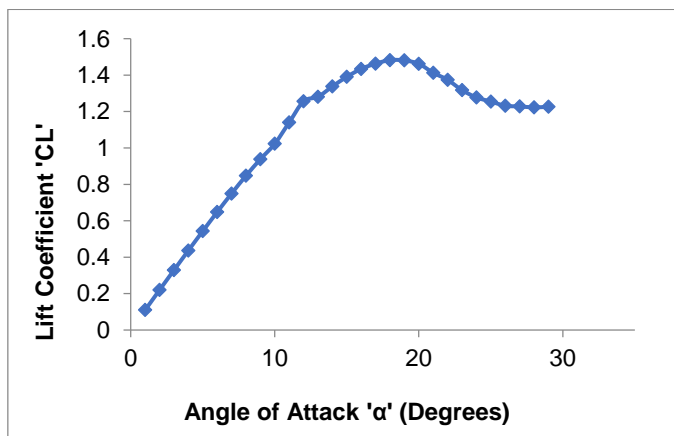
The total thrust and torque obtained from the iteration code written using MATLAB are:

Total Thrust 'T' = 2.2750 × 10<sup>3</sup> N

Total Torque 'Q' = 2.3191 × 10<sup>3</sup> Nm.

The power generated 'P<sub>c</sub>' by the rotor is given by:

The graphs below show the variation of angle of attack 'α' with lift coefficient 'C<sub>L</sub>' and drag coefficient 'C<sub>D</sub>' :



**Figure 2 Variation of lift coefficient with respect to angle of attack**

Power generated ' $P'_G$ ' = Total Torque ' $Q$ '  $\times$  angular velocity of the rotor ' $\Omega$ '

Power generated, ' $P'_G = 2.3191 \times 10^3 N/m \times 6 \text{ rad/s} = 13914.6 \text{ Watts}$

To calculate the power coefficient, ' $C_p$ ' =  $\frac{\text{Power Generated}}{\text{Power Available}} = \frac{P_G}{\frac{1}{2} \rho A U \infty^3}$

Where A = Cross Sectional Area,

$$A = \pi R^2 = \pi \times 6^2 = 36\pi m^2 = 113.1 m^2$$

$$\text{Power Coefficient } 'C_p' = \frac{13914.6}{\frac{1}{2} \times 1.2 \times 113.1 \times 8^3} = \frac{13914.6}{34744.3} = 0.40$$

The value of ' $C_p$ ' obtained above indicates that about forty percent of the kinetic energy of the undisturbed wind is extracted as useful power.

### 1.4 Effect of Blade Twist on Rotor Performance

Twisting of a rotor blade affects the angle of attack ' $\alpha$ ' and the lift to drag ratio  $\frac{F_L}{F_D}$  [5]. Continuous twist of blade along its length helps to achieve optimum angle of attack across the blade's cross-section. As a result of this, the lift is maximized (i.e. high lift to drag ratio is obtained), thus increasing rotational moment and the power output of the turbine rotor [5], [13].

$$\frac{\partial \phi(x,y,t)}{\partial t} = -a \frac{\omega}{k} e^{ky} \times -\omega \times -\sin(kx - \omega t)$$

$$\frac{\partial \phi(x,y,t)}{\partial t} = -a \frac{\omega^2}{k} e^{ky} \sin(kx - \omega t)$$

Therefore, the dynamic pressure is given by the expression:

$$P_d(x,y,t) = -\rho \left[ -a \frac{\omega^2}{k} e^{ky} \sin(kx - \omega t) \right]$$

$$\text{Dynamic Pressure, } P_d(x,y,t) = \rho a \frac{\omega^2}{k} e^{ky} \sin(kx - \omega t)$$

In order to get the final expression for the dynamic pressure, the unknown parameters in the above expression need to be found as shown below:

$$\text{Amplitude 'a'} = \frac{H_w}{2} = \frac{7}{2} = 3.5m,$$

$$\text{Wave Period } 'T_w' = \frac{2\pi}{\omega_w} = \frac{2\pi}{0.6} = 10.47s$$

$$\text{Wavelength } '\lambda' = 1.56 \times T_w^2 = 1.56 \times 10.47^2 = 171m$$

$$\text{Wave number 'k'} = \frac{2\pi}{\lambda} = \frac{2 \times 3.142}{171} = 0.037 \text{ rad/m}$$

The final expression for the dynamic pressure is:

$$P_d(x,y,t) = 1025 \times 10.47 \times \frac{0.6^2}{0.037} \times e^{0.037y} \times \sin(0.037x - 0.6t)$$

$$P_d(x,y,t) = 104417 e^{0.037y} \sin(0.037x - 0.6t) \text{ N/m}^2.$$

### 2.1 Heave Equation of Motion

The heave equation of motion (time domain), assuming that at first approximation the heave degree of freedom is decoupled from the other degree of freedom is given by:

$$(m + A) z'' + Cz = F(t)$$

Where m= mass of the submersible, A= added mass

C= hydrostatic stiffness.

Since the body is a freely floating body, Archimedes' principle can be applied:

Weight of fluid displaced = Weight of the floating body

$\rho V g = mg$ . Therefore, mass of the body 'm' can be obtained from:

Mass of floating body =  $\rho \times V$

where displaced volume 'V' = submerged volume =  $2\pi r^2 T$

## 2 ESTIMATION OF THE TOTAL VERTICAL FORCE ACTING ON A PLATFORM DUE TO THE DYNAMIC FORCE ONLY

The following parameters were given to solve the questions in this chapter:

Diameter of each of the two submerged cylinders 'D' = 10m.

Draft 'T' = 20m,

Width of submersible platform 'L' = 54m

Seawater density ' $\rho$ ' =  $1025 \text{ kg/m}^3$ , Gravitational acceleration ' $g$ ' =  $9.81 \text{ m/s}^2$

Monochromatic wave frequency ' $\omega_w$ ' =  $0.6 \text{ rad/s}$

Wave height ' $H_w$ ' = 7m.

In addition, deep water condition is assumed to be valid.

From Airy wave theory, the velocity potential expression is given by:

$$\phi = -a \frac{\omega}{k} e^{ky} \cos(kx - \omega t) \text{ for deep water condition}$$

Recall dynamic pressure ' $P'_d$ ' is given by:

$$P_d(x,y,t) = -\rho \frac{\partial \phi(x,y,t)}{\partial t}$$

Differentiating the expression for ' $\phi$ ' partially with respect to 't', we have:

$$\frac{\partial \phi(x,y,t)}{\partial t} = \frac{\partial}{\partial t} \left[ -a \frac{\omega}{k} e^{ky} \cos(kx - \omega t) \right]$$

Displaced volume 'V' =  $2 \times \pi \times 5^2 \times 20 = 3141.6m^3$

Mass of floating body 'm' =  $1025 \times 3141.6 = 3220140kg/m^3 = 3.22 \times 10^6 kg/m^3$

The added mass coefficient ' $C_a$ ' chosen from the appendix of [21]DNV RP (2014) by considering the vertical motion of the body and its body shape (chosen as circular disc) is:

Added mass coefficient ' $C_a$ ' =  $(\frac{2}{\pi}) = 0.64$

To calculate the restoring coefficient 'C', we have:

Restoring coefficient 'C' =  $2\pi\rho gr^2 = 2 \times \pi \times 1025 \times 9.81 \times 5^2 = 1579475N/m$

The heave undamped natural frequency of the submersible can be estimated as shown below:

Undamped natural frequency ' $\omega_0$ ' =  $\sqrt{\frac{C}{m+A}}$

Undamped natural frequency ' $\omega_0$ ' =  $\sqrt{\frac{1579475}{3220140+A}}$

Where added mass 'A' =  $2 \times C_a \times \rho \times V_R = 2 \times 0.64 \times 1025 \times \frac{4}{3} \pi \times 5^3$

(Reference Volume =  $\frac{4}{3} \pi R^3$ )

Added mass 'A' = 687050.7kg

Thus, Undamped natural frequency becomes:

Undamped natural frequency,

$$\omega_0 = \sqrt{\frac{1579475}{3220140+687050.7}} = \sqrt{\frac{1579475}{3907190.7}}$$

$$\omega_0 = 0.64rad/s.$$

## 2.2 Natural Frequency in Heave Versus Typical Wave Energy Spectra

The following parameters were substituted into the Pierson-Moskowitz  $S_{PM}(\omega)$  and JONSWAP wave spectrum  $S_j(\omega)$  with the value of frequency ' $\omega$ ' from 0 to 4.0:  $\gamma=3.3$ ,  $H=7m$ ,  $\omega_p=0.6rad/s$ ,  $\sigma=0.09$  (since  $\omega > \omega_p$ ),

$$A=1-\ln \gamma=1-\ln(3.3) = 0.657344252.$$

The varied values of ' $\omega$ ' are plotted against the corresponding values of PM spectrum and JONSWAP spectrum. The natural

frequency in heave ' $\omega_0$ ' of the platform coincides with the highest energy spectrum for both Pierson-Moskowitz  $S_{PM}(\omega)$  and JONSWAP wave spectrum  $S_j(\omega)$ . Also, it is clearly seen from the graph below that JONSWAP wave spectrum  $S_j(\omega)$  has more energy than Pierson-Moskowitz  $S_{PM}(\omega)$  wave spectrum at the natural frequency (i.e.  $\omega_0 = 0.64rad/s$ ) in heave of the platform obtained from the previous section.

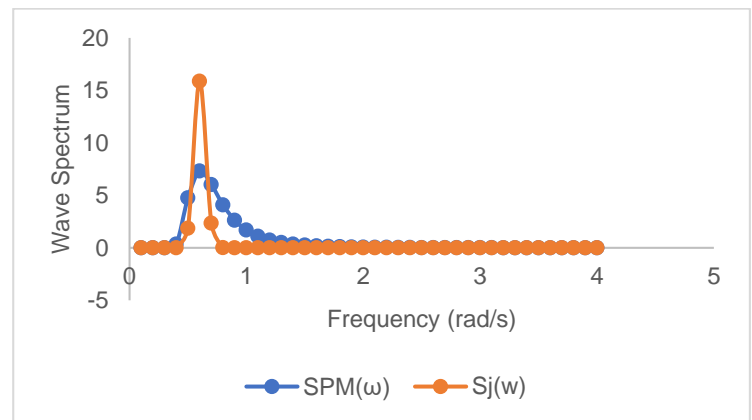


Figure 5 Wave Energy Spectra

## 3 ESTIMATION OF HIGHEST SIGNIFICANT WAVE HEIGHT USING WEIBULL PDF.

Given parameters include:

Weibull fitting parameter 'a'=0.5

Added mass coefficient ' $C_a$ ' = 1.0

Drag Coefficient ' $C_d$ ' = 0.8

Gravitational acceleration ' $g$ ' = 9.8m/s<sup>2</sup>

Highest wave period ' $T_{Hmax}$ ' = 13s

Diameter 'D' = 3m

Length of member 'L' = 15m

Draft 'F' = 75

Return period 'R' = 1 year

Seawater density ' $\rho$ ' = 1025kg/m<sup>3</sup>

Seawater Viscosity ' $\nu$ ' =  $1.05 \times 10^{-6} m^2/s$

Water depth 'd' = 200m

JONSWAP, peak frequency ' $\omega_p$ ' = 0.45rad/s

JONSWAP, peak shape par ' $\gamma$ ' = 3.3

Table 2 Weibull Distribution Data

H/m	$P_{CL}(H)$	$\ln(H - 0.5)$	$\ln[\ln(\frac{1}{1 - P_{CL}(H)})]$
1	0.074	-0.693	-2.565
2	0.336	0.405	-0.893
3	0.629	0.916	-0.008
4	0.825	1.253	0.556
5	0.925	1.504	0.952
6	0.968	1.705	1.236

7	0.986	1.872	1.451
8	0.994	2.015	1.632
9	0.998	2.140	1.827
10	10	2.251	0

The probability that the wave will be lower than the 1year return wave:  
 $P_1(H) = P(H < H_1) = 1 -$

$$\frac{3h}{24 \frac{h}{d} \times 365.25 \frac{d}{y} \times 1y} \approx 0.9997$$

Thus;

$$\ln[\ln(\frac{1}{1-P_1(H)})] = \ln[\ln(\frac{1}{1-0.9997})] = 2.093$$

With the table above, a Weibull plot was created as shown below:

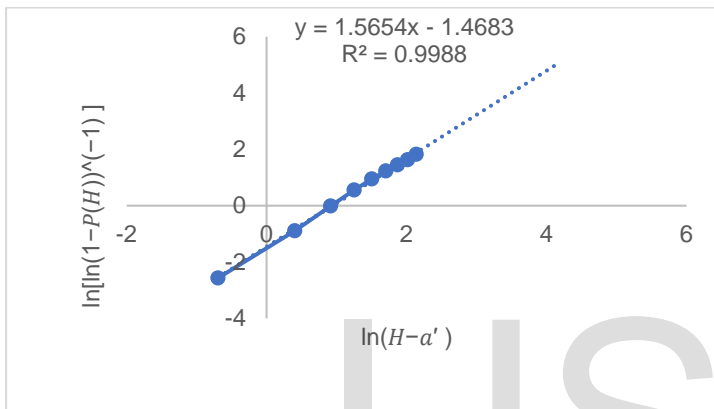


Figure 6 Weibull Plot

It can be seen from the Weibull plot that the equation of the line is:

$$Y = 1.5654x - 1.4683 \text{ and } \text{Correlation Coefficient "R"} = 0.9988.$$

The value of 'R' shows that a very strong relationship exists between the values plotted, as it is close to 1.0. To know the corresponding value of  $\ln(1 - a')$  (i.e. value on x-axis) when  $\ln[\ln(1 - P(H))^{-1}] = 2.093$ , we use the equation of the line described above:

$$\begin{aligned} 2.093 &= 1.5654x - 1.4683 \\ 1.5654x &= 2.093 + 1.4683 \\ 1.5654x &= 3.5613. \end{aligned}$$

$$\text{Thus, } x = \frac{3.5613}{1.5654} = 2.275$$

Therefore, the corresponding value of  $\ln(1 - a')$  when  $\ln[\ln(1 - P(H))^{-1}] = 2.093$  is 2.275. Thus, the maximum significant wave height for the 1year return period is given by:

$$\begin{aligned} \ln(H_1 - a') &= 2.275 \\ H_1 &= e^{2.275} + a' = 9.73 + 0.5 \text{ (where } a' = 0.5) \\ H_1 &= 10.23m \approx 10m \end{aligned}$$

Therefore, the maximum significant wave height for one year return period is 10m.

### 3.1 Calculation of Highest Wave Height likely to occur in One Year Return Period.

The parameters given to calculate the highest wave height likely to occur are:

JONSWAP, peak frequency ' $\omega_p$ ' = 0.45rad/s

JONSWAP, peak shape par ' $\gamma$ ' = 3.3

$$\text{Peak frequency } \omega_p' = \frac{2\pi}{\text{JONSWAP, Peak period}} = \frac{2\pi}{T_p}$$

$$\text{Thus, JONSWAP, peak period } T_p' = \frac{2\pi}{\omega_p} = \frac{2\pi}{0.45} = \frac{2 \times 3.142}{0.45} = \frac{6.284}{0.45} = 13.96s$$

In section 3.5.5 of DNV-RP-C205[21], JONSWAP, peak period ' $T_p$ ' is approximated as:

$$T_p = 1.2859T_z, \text{ when } \gamma = 3.3$$

$$T_z = \frac{T_p}{1.2859} = \frac{13.96}{1.2859} = 10.86s$$

Thus, the zero up-crossing wave period,  $T_z$  is 10.86s.

Thus, to calculate the mean wave period  $T_1 = 1.0734T_z = 1.0734 \times 10.86$

Mean wave period  $T_1 = 11.66s$ .

Therefore, the total number of waves ' $N_{waves}$ ' to be encountered in 3 hours is given by:

$$N_{waves} = \frac{t}{T_1} = \frac{3 \times 3600}{11.66} = 926.24 \text{ waves}$$

$$N_{waves} \approx 926 \text{ waves will be encountered}$$

The probability of obtaining the highest wave likely to occur is:

$$\begin{aligned} P(H_{max}) &= \frac{N_{waves} - 1}{N_{waves}} = \frac{926 - 1}{926} \\ P(H_{max}) &= \frac{925}{926} = 0.999 \end{aligned}$$

The highest wave likely to occur in the given return period,  $H_{max}$  is given by:

$$H_{max} = \sqrt{\ln(1 - P(H_{max}))} \times \frac{H_{1/3}^2}{-2} \text{ where } H_{1/3} = 10.23m$$

$$H_{max} = \sqrt{\ln(1 - 0.999)} \times \frac{10.23^2}{-2}$$

$$H_{max} = \sqrt{361.458} = 19.01m \approx 19m.$$

Therefore, the highest wave likely to occur in the given return period is 19m.

### 3.2 Selection of Appropriate Wave Theory

In order to select the appropriate wave theory, we need to calculate the following parameters:

1.  $\frac{d}{\lambda}$ : To calculate the wavelength ' $\lambda$ ' of the wave we use:

$$\lambda = 1.56 \times T_{Hmax}^2 \text{ where } T_{Hmax} = 13s.$$

$$\lambda = 1.56 \times 13^2 = 1.56 \times 169$$

$$\lambda = 263.64 \text{m.}$$

Given depth of water 'd'=200m:  $\frac{d}{\lambda} = \frac{200}{263.64} \approx 0.76$

Since  $\frac{d}{\lambda} \gg \frac{1}{20}$  (i.e.  $0.76 \gg 0.05$ ), therefore we assume deep-water condition.

2.  $\frac{H}{gT^2}$ : To calculate this, we use the values of  $H_{max}$  and  $T_{Hmax}$  as stated in the question:

$$\frac{H_{max}}{gT^2_{Hmax}} = \frac{19}{9.81 \times 13^2} = \frac{19}{9.81 \times 169} = \frac{19}{1657.89} = 0.0115.$$

Where acceleration due to gravity 'g' = 9.81m/s<sup>2</sup>

3.  $\frac{d}{gT^2}$ : To calculate this, we use the values of 'd' and  $T_{Hmax}$  as stated in the question:

$$\frac{d}{gT^2_{Hmax}} = \frac{200}{9.81 \times 13^2} = \frac{200}{9.81 \times 169} = \frac{200}{1657.89} = 0.121.$$

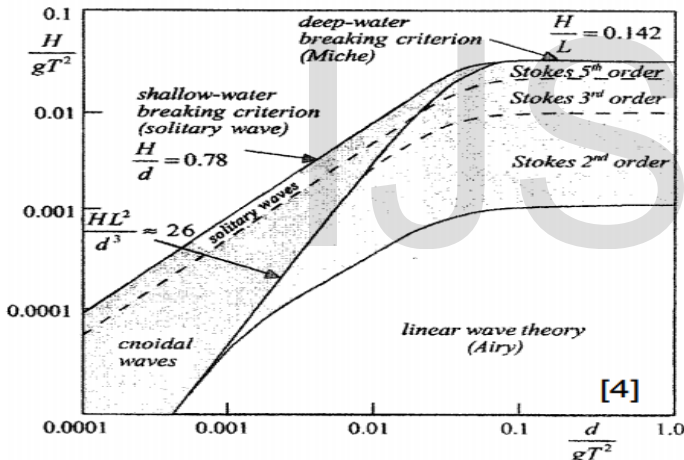


Figure 7 Graph of range of validity for wave theories [22].

From the figure above, by tracing the point 0.0115 on the vertical axis and 0.121 on the horizontal axis, it is seen that the point of intersection of these two points in the Stoke's 3<sup>rd</sup> order region of the graph. Thus, Stoke's 3<sup>rd</sup> order should be used.

### 3.3 Estimation of Wave Force Using Morison Approach

In order to demonstrate that the wave force can be estimated using Morison approach [23], [24], we need to calculate the parameters listed below:

1.  $\frac{\text{Member Diameter}}{\text{Wavelength}} = \frac{D}{\lambda}$

If  $\frac{D}{\lambda} < 0.2$ , then the characteristics of the incident wave is not significantly altered (i.e. a small volume structure), therefore Morison equation will be applicable. Thus: Where Member Diameter 'D' = 3m

Wavelength ' $\lambda$ ' = 263.64m (as calculated in section 3.3)

Therefore:  $\frac{D}{\lambda} = \frac{3}{263.64} = 0.0114$

Since  $0.0114 \ll 0.2$ , Morison equation is applicable to estimate the wave force.

2. Keulegan-Carpenter number ' $K_C$ ' =  $\frac{U_m \times T_{Hmax}}{D}$

Where  $U_m$  = Maximum normal water particle velocity, D = Member Diameter = 3m,  $T_{Hmax}$  = Highest wave Period = 13s.

The value of ' $U_m$ ' can be determined from Airy wave theory velocity potential as shown below:

$$\phi = -a \frac{\omega}{K} e^{ky} \cos(kx - \omega t)$$

The above equation holds for deep water condition.

The vertical velocity is given by:

$$U = \frac{\partial \phi}{\partial x} = \frac{\partial}{\partial x} \left[ -a \frac{\omega}{K} e^{ky} \cos(kx - \omega t) \right]$$

$$U = a \omega e^{ky} \sin(kx - \omega t)$$

Therefore, the maximum horizontal velocity ' $U'_m$ ' occurs at waterline level (i.e.  $y=0$ ) and when  $(kx - \omega t)$  equals  $90^\circ$ , thus:

$$U_m = a \omega = \frac{H_{max}}{2} \times \frac{2\pi}{T_{Hmax}} = \frac{\pi H_{max}}{T_{Hmax}}$$

$$U_m = \frac{3.142 \times 19}{13} = 4.59 \text{m/s.}$$

Now, the Keulegan-Carpenter number ' $K_C$ ' can be estimated as:

$$K_C = \frac{4.59 \times 13}{3} = 19.89$$

Since  $5 < K_C < 25$  (i.e.  $5 < 19.89 < 25$ ), only Morison equation is applicable to estimate the wave force. The table below explains this better, as this situation falls in the intermediate regime based on the values obtained above:

	D/A < 0.20	D/A > 0.20
K-C > 25 (H/D > 8) (Deepwater K-C~πH/D)	<b>DRAG DOMINATED</b> <u>Morison Eqn.</u> C <sub>M</sub> = 1.8 C <sub>D</sub> function of Re	(no waves here, above breaking condition)
5 < K-C < 25 (1.5 < H/D < 8)	<b>INTERMEDIATE REGIME</b> Complex flow, exp data wide scatter <u>Morison Eqn</u> For Re > 1.5E06 C <sub>M</sub> = 1.8 C <sub>D</sub> = 0.62	(no waves here, above breaking condition)
K-C < 5 (H/D < 1.5)	<b>INERTIA DOMINATED</b> <u>Morison Eqn / Diffraction Theory</u> C <sub>M</sub> = 2.0 C <sub>D</sub> = negligible	<u>NO Morison Eqn</u> <u>Diffraction Theory</u>

Figure 8 Wave Loading Regimes [25].

To derive the expression for the total vertical force due to the wave acting on the offshore structure [26]–[30] (i.e. a horizontal cylinder normal to the wave), we have:

Total Vertical Force  $F_V =$  Inertia Force ( $F_I$ ) + Drag force ( $F_D$ )

$$F_V = \int_{-L/2}^{L/2} (C_M \rho \pi r^2 \dot{V} dx + C_D \rho r |V| V dx)$$

Where  $C_M =$  Inertia Coefficient = added mass coefficient + 1 =  $C_a + 1$

Given added mass coefficient  $C_a = 1$ ; Thus  $C_M = 1+1 = 2$

Seawater density ' $\rho$ ' = 1025kg/m<sup>3</sup>, Radius ' $r$ ' =  $\frac{D}{2} = \frac{3}{2} = 1.5m$

Length ' $L$ ' = 15m Drag coefficient ' $C_D$ ' = 0.8

The values of vertical acceleration ' $\dot{V}$ ' and vertical velocity ' $V$ ' can be calculated using Airy wave theory velocity potential as shown below:

$$\phi = -a \frac{\omega}{K} e^{ky} \cos(kx - \omega t) \text{ for deep water condition}$$

$$\text{Vertical velocity 'V'} = \frac{\partial \phi}{\partial y} = \frac{\partial}{\partial y} \left[ -a \frac{\omega}{K} e^{ky} \cos(kx - \omega t) \right]$$

$$\text{Vertical velocity 'V'} = -a \omega e^{ky} \cos(kx - \omega t)$$

$$\text{Vertical acceleration '}\dot{V}\text{' } = \frac{\partial V}{\partial t} = \frac{\partial}{\partial t} [-a \omega e^{ky} \cos(kx - \omega t)]$$

$$\text{Vertical acceleration '}\dot{V}\text{' } = -a \omega^2 e^{ky} \sin(kx - \omega t)$$

The values of wave number ' $k$ ', amplitude ' $a$ ' and circular frequency ' $\omega$ ' are calculated as shown below:

$$\text{Wave number 'k'} = \frac{2\pi}{\lambda} = \frac{2 \times 3.142}{263.64} = \frac{6.284}{263.64} \approx 0.024 \text{ rad/m,}$$

$$\text{Amplitude 'a'} = \frac{H_{max}}{2} = \frac{19}{2} = 9.5m,$$

$$\text{Circular Frequency '}\omega\text{' } = \frac{2\pi}{T_{Hmax}} = \frac{2 \times 3.142}{13} = \frac{6.284}{13} \approx 0.48 \text{ rad/s.}$$

Therefore, the expressions for the vertical velocity ' $V$ ' and vertical acceleration ' $\dot{V}$ ' are shown below:

$$\text{Vertical velocity 'V'} = -9.5 \times 0.48 \times e^{0.024y} \times \cos(0.024x - 0.48t)$$

$$\text{'V'} = -4.56 e^{0.024y} \cos(0.024x - 0.48t)$$

$$\text{Vertical acceleration '}\dot{V}\text{' } = -9.5 \times 0.48^2 \times e^{0.024y} \times \sin(0.024x - 0.48t)$$

$$\text{'}\dot{V}\text{' } = -2.189 e^{0.024y} \sin(0.024x - 0.48t)$$

The total vertical force becomes:

$$\text{Total vertical force } F_V = \int_{-7.5}^{7.5} (C_M \rho \pi r^2 \dot{V} dx + C_D \rho r |V| V dx)$$

$$\text{Vertical Inertia force } F_I = \int_{-7.5}^{7.5} C_M \rho \pi r^2 \dot{V} dx$$

$$F_I = \int_{-7.5}^{7.5} [2 \times 1025 \times 3.142 \times 1.5^2 \times -2.189 e^{0.024y} \sin(0.024x - 0.48t) dx]$$

$$F_I = \int_{-7.5}^{7.5} [-31724.03 e^{0.024y} \sin(0.024x - 0.48t) dx]$$

$$F_I = -31724.03 e^{0.024y} \int_{-7.5}^{7.5} [\sin(0.024x - 0.48t) dx]$$

$$F_I = 31724.03 e^{0.024y} \left[ \frac{\cos(0.024x - 0.48t)}{0.024} \right]_{-7.5}^{7.5} = \frac{31724.03}{0.024} e^{0.024y} [\cos(0.024x - 0.48t)]_{-7.5}^{7.5}$$

$$F_I = 1321834.6 e^{0.024y} [\cos((0.024 \times 7.5) - 0.48t) - \cos((0.024 \times -7.5) - 0.48t)]$$

$$F_I = 1321834.6 e^{0.024y} [\cos(0.18 - 0.48t) - \cos(-0.18 - 0.48t)]$$

$$F_I = 1321834.6 e^{0.024y} [\cos(0.18 - 0.48t) - \cos(-0.18 - 0.48t)]$$

Taking  $y=0$

$$F_I = 1321834.6 [\cos(0.18 - 0.48t) - \cos(-0.18 - 0.48t)]$$

Therefore, the inertia force component ' $F_I$ ' of the vertical force is  $1321834.6 [\cos(0.18 - 0.48t) - \cos(-0.18 - 0.48t)]$ .

To calculate the Drag force component,  $F_D$ , of the total vertical force, we have:



$$F_D = \int_{-7.5}^{7.5} C_D \rho r |V| V dx$$

$$F_D = \int_{-7.5}^{7.5} [0.8 \times 1025 \times 1.5 \times | -4.56e^{0.024y} \cos(0.024x - 0.48t) | \times (-4.56e^{0.024y} \cos(0.024x - 0.48t))] dx$$

$$F_D = 25576.13 \int_{-7.5}^{7.5} [e^{0.024y} \cos(0.024x - 0.48t) | \times (e^{0.024y} \cos(0.024x - 0.48t))] dx$$

Taking  $y=0$ , the above integral for drag force component ' $F_D$ ' is evaluated using MATLAB software and the result is shown below:

$$F_D = 532836.04 \sin(0.096t) - 266418 \sin(0.096t - 0.36) - 266418 \sin(0.096t + 0.36)$$

The total vertical force acting on the offshore structure is:

$$F_V = F_I + F_D$$

The total vertical force acting on the offshore structure is:

$$F_V = 1321834.6e^{0.024y} [\cos(0.18 - 0.48t) - \cos(-0.18 - 0.48t)] + 532836.04 \sin(0.096t) - 266418 \sin(0.096t - 0.36) - 266418 \sin(0.096t + 0.36)$$

The figure below shows how inertia force ' $F_I$ ', drag force ' $F_D$ ' and total vertical force ' $F_V$ ' vary with time:

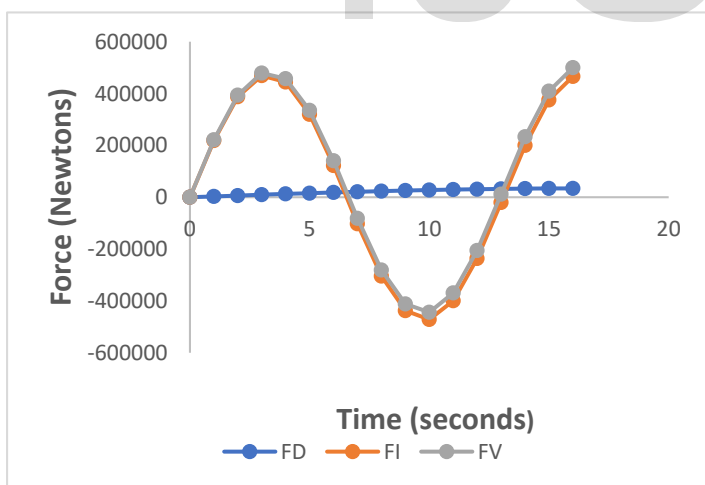


Figure 9 Variation of Force with Time

### 3.4 Limitations of Approaches Adopted in Previous Steps

Weibull distribution was the approach used in section 3.1 and it can only be used to estimate long-term wave properties, usually 10 years and above[22]. It is obvious that there are no data measurements covering a period of at least 100 years properties, thus estimates rely on extrapolation of wave

properties measured by devices over a period of 3 years[22]. In conclusion, Weibull distribution is prone to error as every estimate depends on extrapolation[22].

JONSWAP wave spectrum was used in section 3.2 can only be used to analyze the wave properties of young seas [22]. Also, JONSWAP spectrum is applicable to only random unidirectional sea. Stoke's 3<sup>rd</sup> order theory suggested in section 3.3 is only applicable for deep water but it isn't applicable for shallow water, when  $d < \frac{\lambda}{8}$  [31].

Finally, Morison approach used in section 3.4 is not applicable for calculating the wave loads on an offshore structure when the ratio of its diameter (D) to the

wavelength ( $\lambda$ ) of the incident wave is greater than 0.2 (i.e.  $\frac{D}{\lambda} > 0.2$ ) or when the offshore structure is big enough to change the incident wave characteristics [25].

In addition, the partly used Airy linear wave theory in section 3.4 is not applicable for high height waves, that is why it is referred to as small amplitude theory [22][8], [32].

## 4 CONCLUSION

The value of the power coefficient ( $C_p = 0.40$ ) obtained in section one indicates that about forty percent of the kinetic energy of the undisturbed wind that flows to the swept area of the rotor can be extracted as useful power [19]. In addition, the twisting of the blades of a horizontal axis wind turbine has a positive impact on the performance of its rotor, as twisting optimize angle of attack and maximize the lift force [19]. Others factors that improves the performance of a horizontal axis wind turbine are high lift-to-drag ratio, high maximum lift coefficient and high tip speed ratio [19], [33]–[35].

More so, the natural frequency in heave ' $\omega_0$ ' of the platform in section 2 coincides with the highest energy spectrum for both Pierson-Moskowitz  $S_{PM}(\omega)$  and JONSWAP wave spectrum  $S_j(\omega)$ . Also, it is clearly seen from the graph below that JONSWAP wave spectrum  $S_j(\omega)$  has more energy than Pierson-Moskowitz  $S_{PM}(\omega)$  wave spectrum at the natural frequency (i.e.  $\omega_0 = 0.64 \text{ rad/s}$ ) in heave of the platform.

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